Near-Earth Orbit Propagation with Gaia Observations

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The	Gaia	Mission

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2 Astrometry for known NEAs

- Observations of Apophis
- Orbital improvement

3 Astrometry for newly discovered asteroids

- Follow-up network
- NEA alerts expected
- Strategy of recovery

The	Gaia	Mission





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Astrometry for known NEAs

Astrometry for newly discovered asteroids

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The mission



Conclusion

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Conclusion

Nominal scanning law



Fixed parameters

- Basic angle = 106.5°
- Spin rotation = 1° /min
- Spin orientation = 45°
- Spin precession = 63 days

Free parameters

Initial spin phase ⇒ date of observation
 Initial precession angle ⇒ number of observations

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Nominal scanning law







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Observations of Apo	phis		
Precess	sion angle influence	Set used for simulation	n





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Orbital improvement		

Orbital elements

Position uncertainty

	σ (G.B. only)	σ (G.B. + Gaia)
a [AU]	$1.3 imes 10^{-08}$	6.8×10^{-11}
е	$5.7 imes 10^{-08}$	3.9×10^{-09}
i[°]	$1.8 imes 10^{-06}$	1.2×10^{-07}
Ω[°]	$8.0 imes 10^{-05}$	2.2×10^{-06}
ω [°]	$8.0 imes 10^{-05}$	$2.3 imes 10^{-06}$
M [°]	$5.5 imes 10^{-05}$	$6.5 imes 10^{-07}$



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Orbital improvement		

Position uncertainty improvement



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Orbital improvement		00

Conclusion

2029-b-plane uncertainty improvement



2029-b-plane uncertainty

	S ₁	S ₂	S ₃	S ₄	S ₅
σ_{ξ} (km)	10	0.3	7	8	6
σ_{ζ} (km)	240	1.6	10.5	24	11.5

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Future stellar catalogue (\sim 2020)

Orbital elements

2029-b-plane

	Without reduction	With reduction
a [AU]	1.3×10^{-08}	1.3×10^{-09}
е	$5.7 imes 10^{-08}$	$4.9 imes 10^{-09}$
i [°]	1.8×10^{-06}	$9.9 imes 10^{-09}$
Ω[°]	8.0×10^{-05}	$3.9 imes 10^{-06}$
ω [°]	$8.0 imes 10^{-05}$	$3.9 imes 10^{-06}$
M [°]	$5.5 imes 10^{-05}$	$5.6 imes 10^{-06}$

	Without reduction	With reduction
σ_{ξ} (km)	10	1.5
σ_{ζ} (km)	240	23.5

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Outline



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Follow-up network		

Gaia FUN-SSO

Discoveries are expected
Gaia is not a follow-up mission
Track potential alert
Optimisation of the alert-mode



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NEA alerts expected			



Alerts identification

• $H_{min} < H < H_{max}$ • $a_{min} < a < a_{max}$





The	Gaia	Mission

Astrometry for known NEAs

Astrometry for newly discovered asteroids

0 000 00 Conclusion

Strategy of recovery



Method

Statistical Ranging for orbit determination
 Extract Maximum Likelyhood (ML) of the (α,δ) distribution
 Center a given window (FOV_α, FOV_δ) on the ML

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Strategy of recovery

Application for other PHAs



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Strategy of recovery

Synergy ground/space-based data

Geographos recovered on 3rd day
 One night follow-up (8 obs.~ 500 mas)
 ⇒ ∖_x FOV required
 ⇒ optimisation of the data processing



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Outline





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